

Numerical Study on Onset of Turbulence using Interaction between Streaky Structure and Jet

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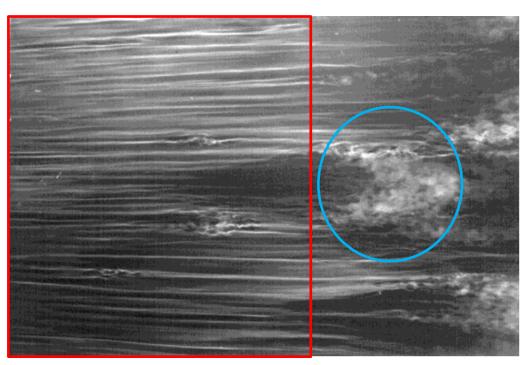
Introduction

Bypass transition (Tu > 1%)

Streaky structures develops in L.B.L.



Turbulent spots randomly appear.



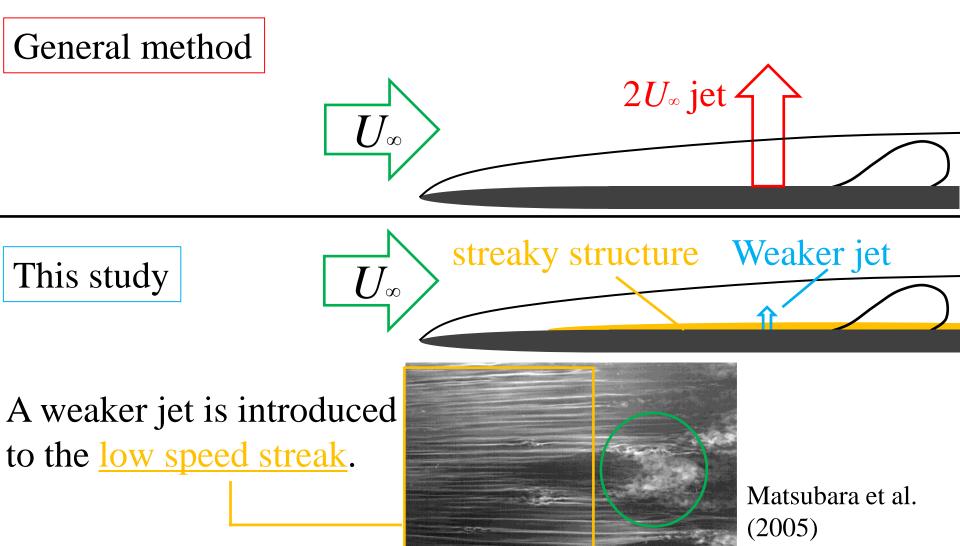
Matsubara et al. (2001)

Turbulent spots fill the B.L.



End of Transition

Turbulent spots could be artifitially generated



Objective

To understand the mechanism that leads to the onset of turbulence.



How the flow field changes downstream of jet is studied using numerical simulation.

Computational method

Governing equations

 3-dimensional incompressible Navier-Stokes equations $\frac{\partial \boldsymbol{u}}{\partial t} + (\boldsymbol{u} \cdot \nabla)\boldsymbol{u} = -\nabla p + \frac{1}{Re}\nabla^2 \boldsymbol{u}$ Continuity equation $\nabla \cdot \boldsymbol{u} = 0$

Computational algorithm

MAC method

Discretization

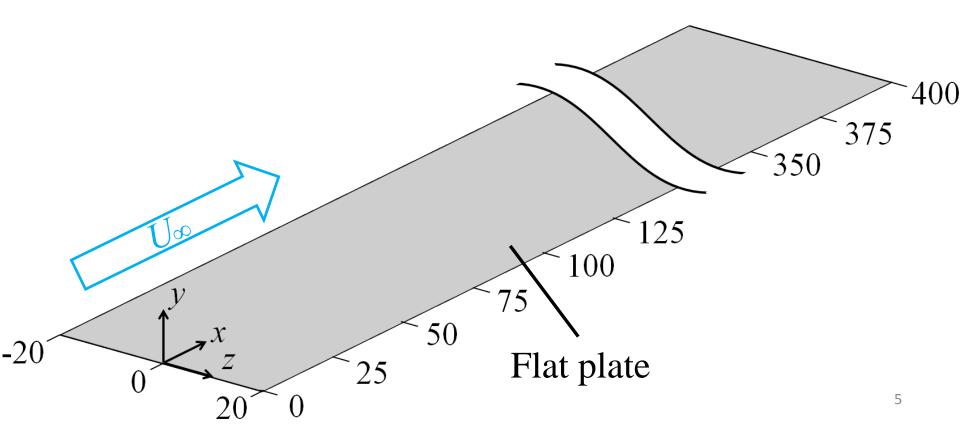
- •Time 2nd-order Crank-Nicolson method
- Space Multi-directional difference scheme
 - □ conv. : 3rd-order upwind difference scheme others : 2nd-order central difference scheme

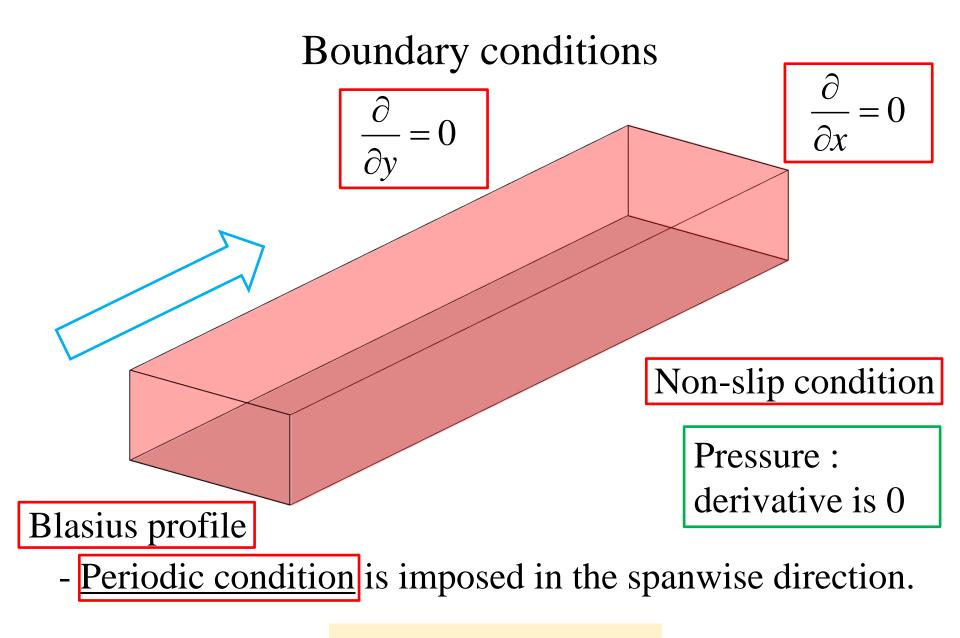
- Only the difference from the Blasius profile is computed.⁴

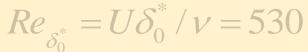
Preliminary computation

Grid number : $2001 \times 101 \times 201$

Characteristic velocity U_{∞} : Uniform velocity Characteristic length δ_0^* : Displacement thickness at inlet

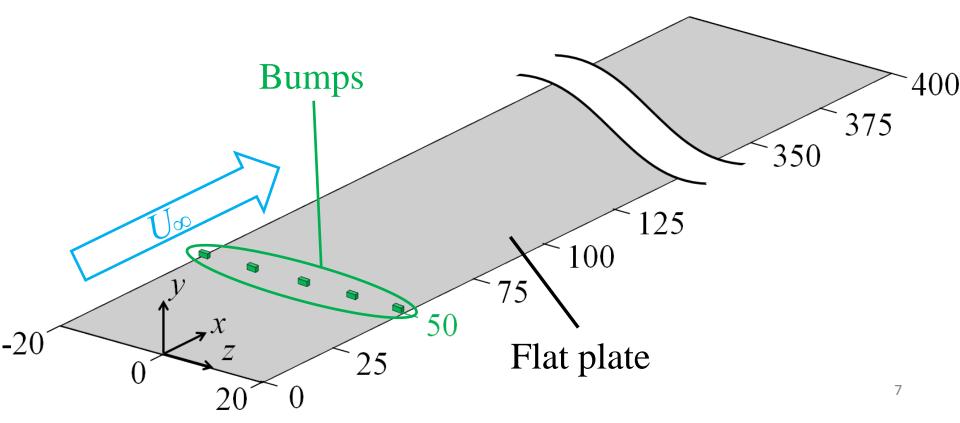






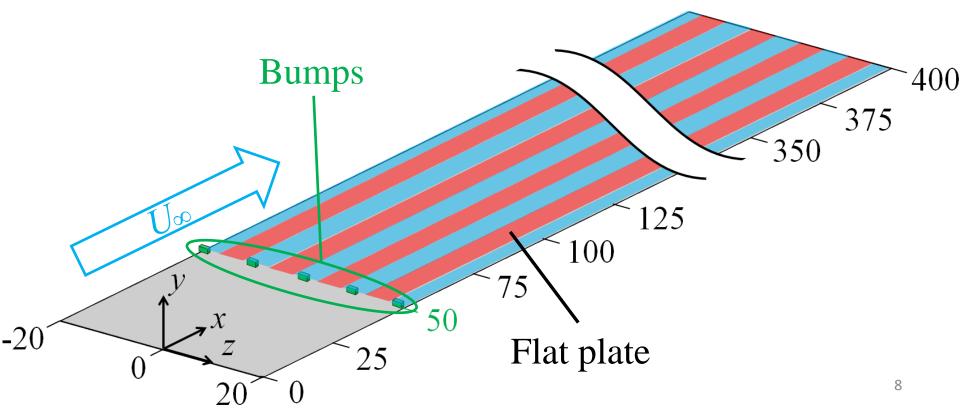
Preliminary computation Grid number : 2001 × 101 × 201

Bumps are installed in a Blasius boundary layer.



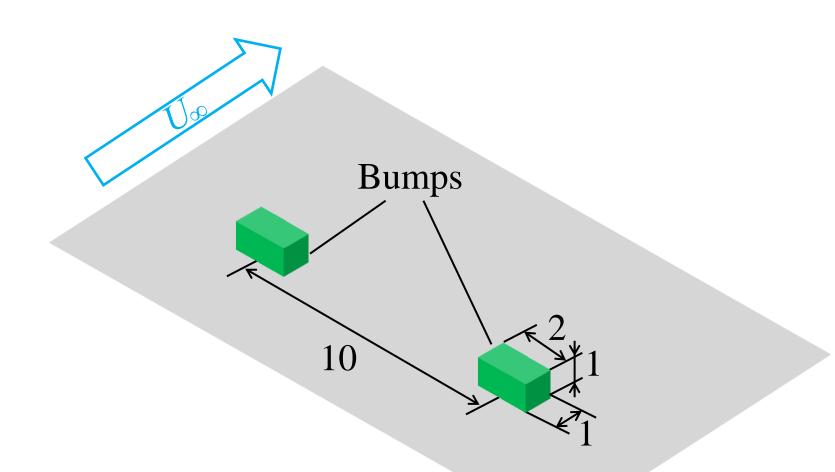
Preliminary computation Grid number : 2001 × 101 × 201

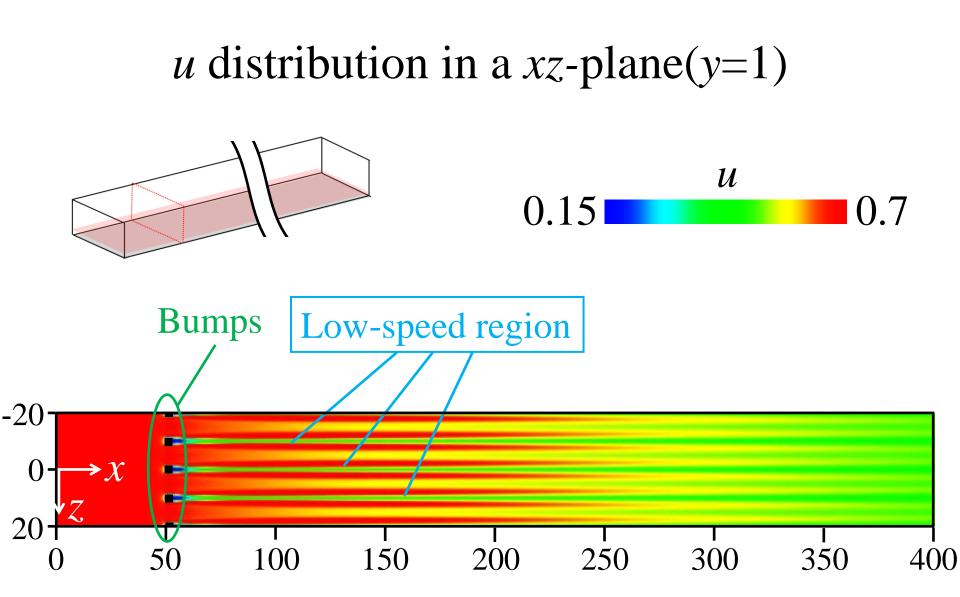
Bumps are installed in a Blasius boundary layer.



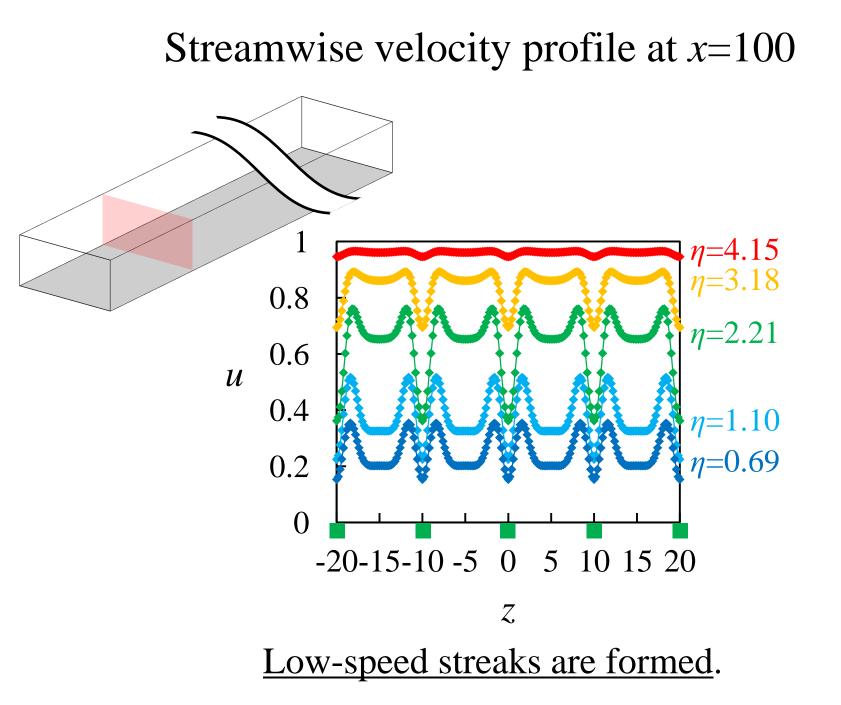
Preliminary computation Grid number : 2001 × 101 × 201

Bumps are installed in a Blasius boundary layer.

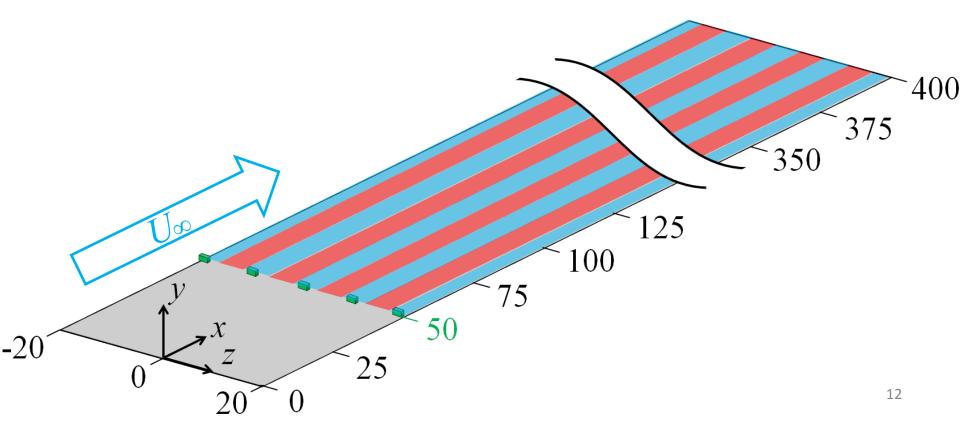




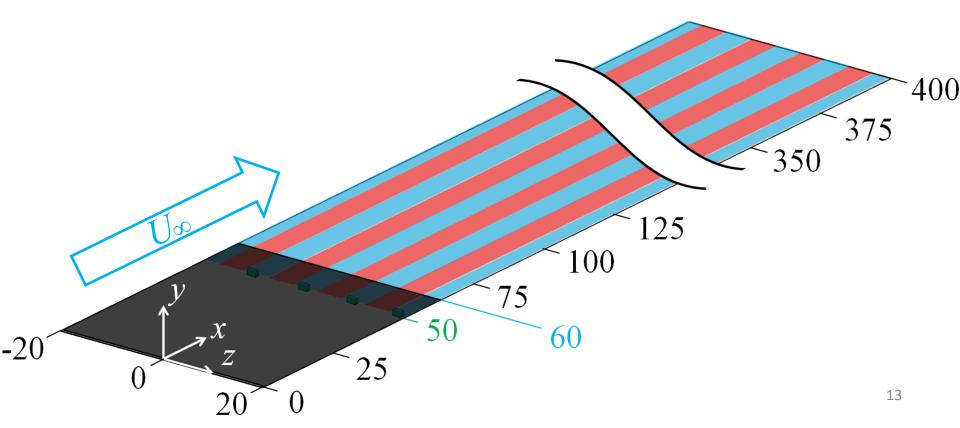
Low-speed streaks are aligned in the spanwise direction.



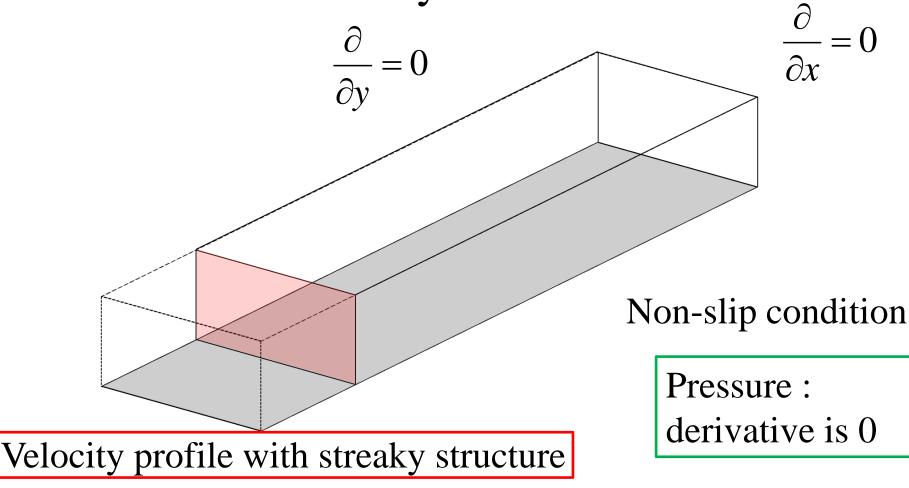
Main computation Grid number : 2001 × 101 × 201



Main computation Grid number : $2001 \times 101 \times 201 \rightarrow 1701 \times 101 \times 201$



Boundary conditions

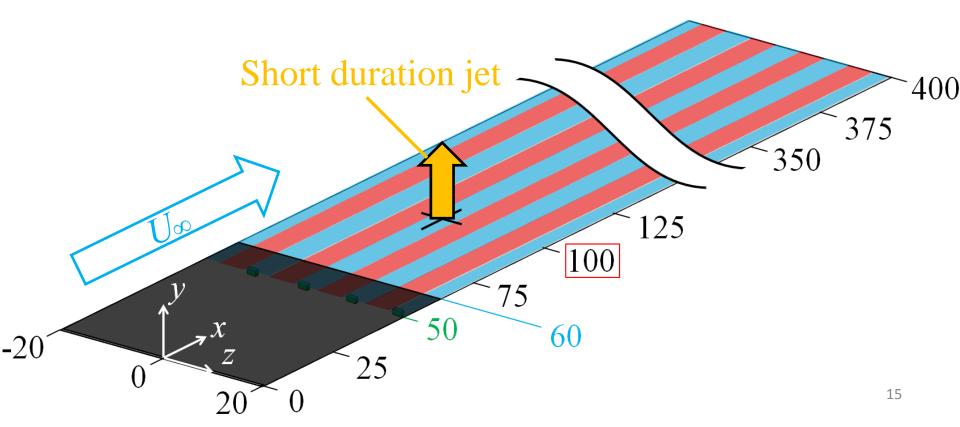


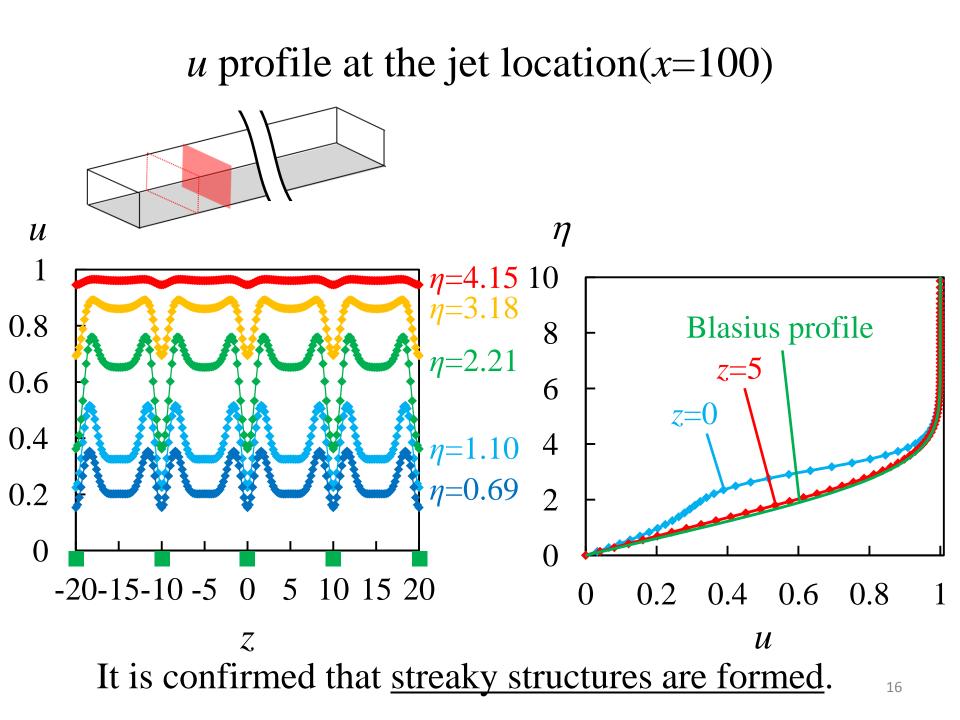
- Periodic condition is imposed in the spanwise direction.

$$Re_{\delta_0^*} = U\delta_0^* / v = 530$$

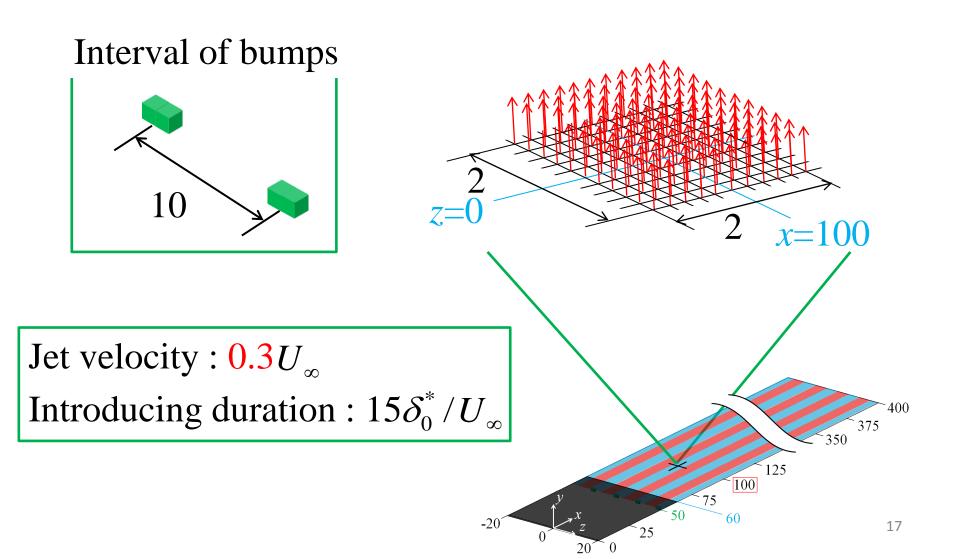
Main computation Grid number : $2001 \times 101 \times 201 \rightarrow 1701 \times 101 \times 201$

Short duration jet is introduced into a low-speed streak.

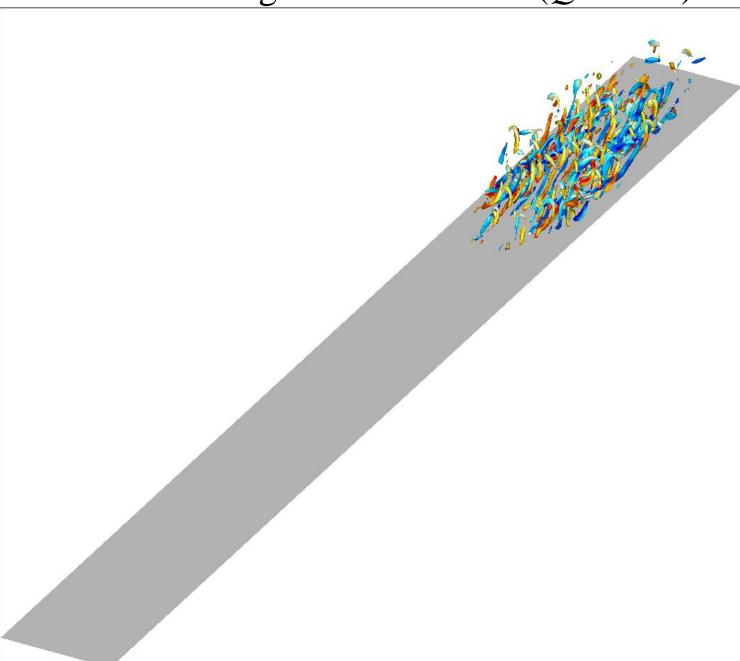


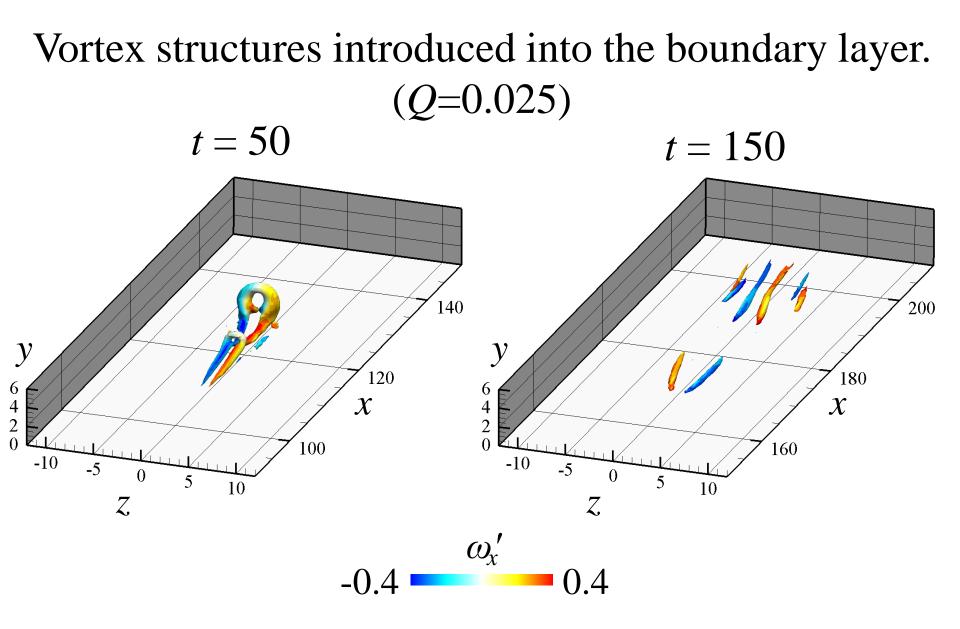


Short duration jet



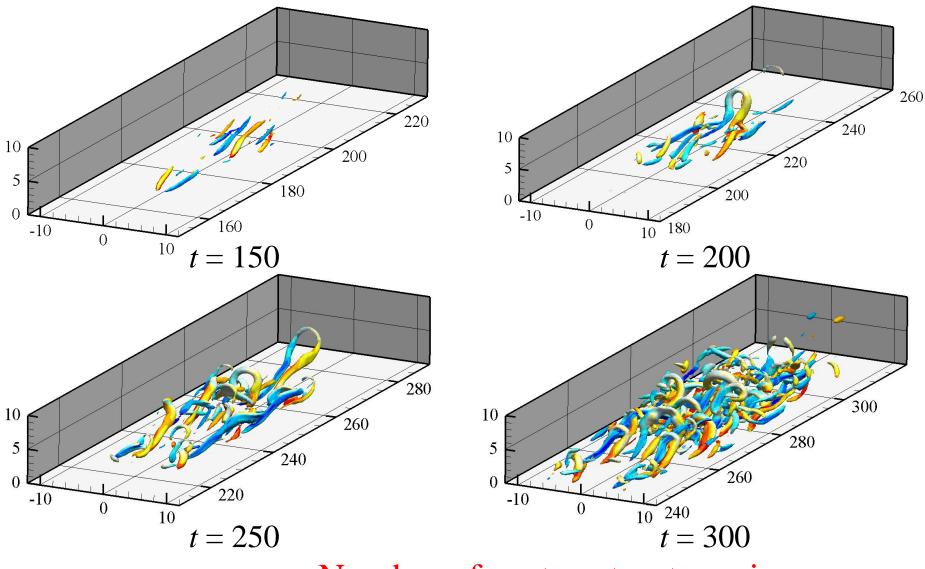
Movie showing vortex structures(*Q*=0.025)





The initial vortex structures decay.

Revival of vortex structures (*Q*=0.025)



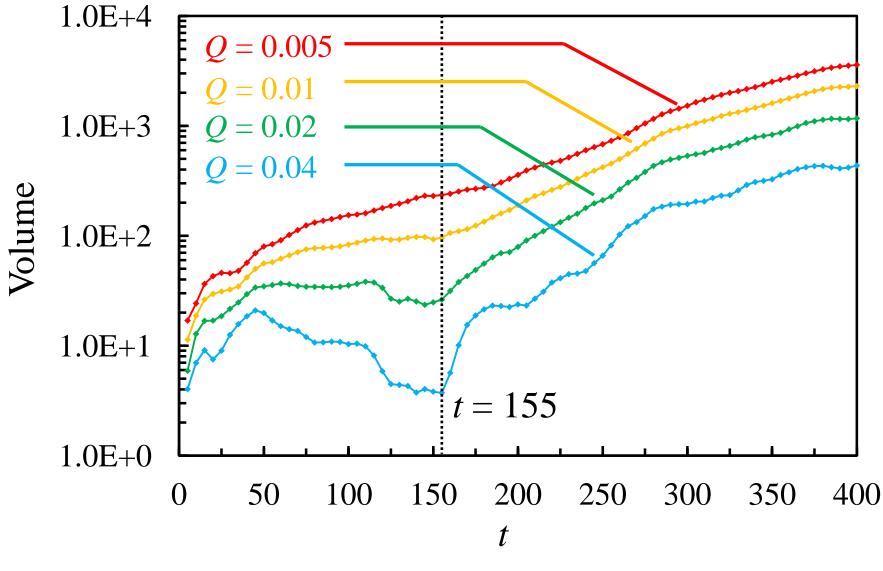
 ω'_x

-0.4

0.4

Number of vortex structures increase.

Change in total volume of vortex structures

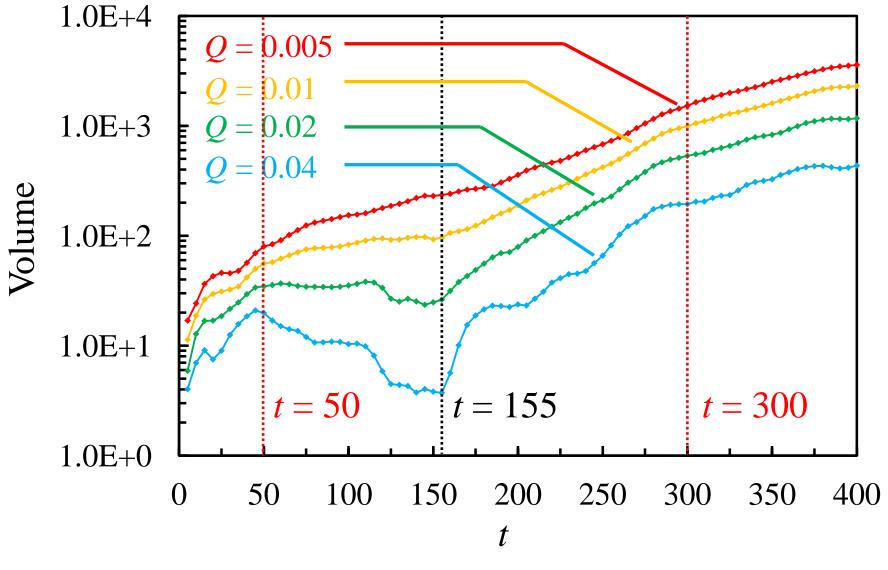


Turbulence starts around t=155.

Under what condition does the turbulence start?

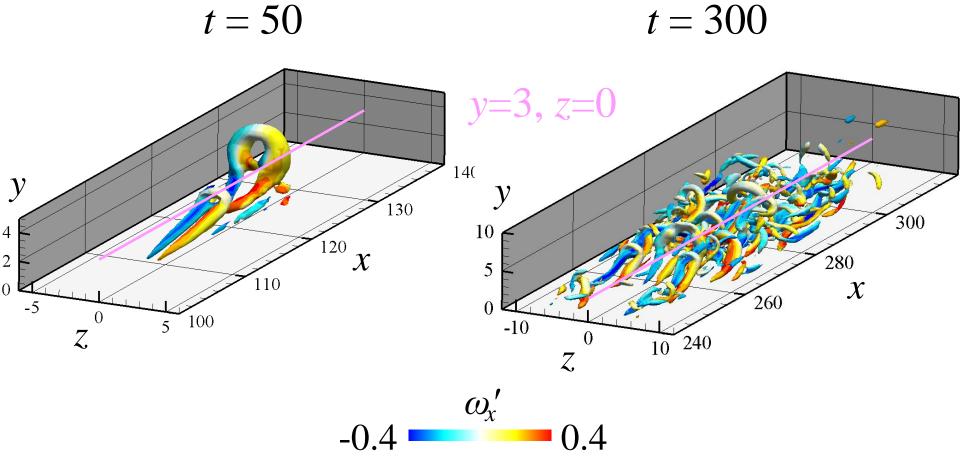
Any good way to identify the onset of turbulence ?

Time variation of total volume of vortex structures



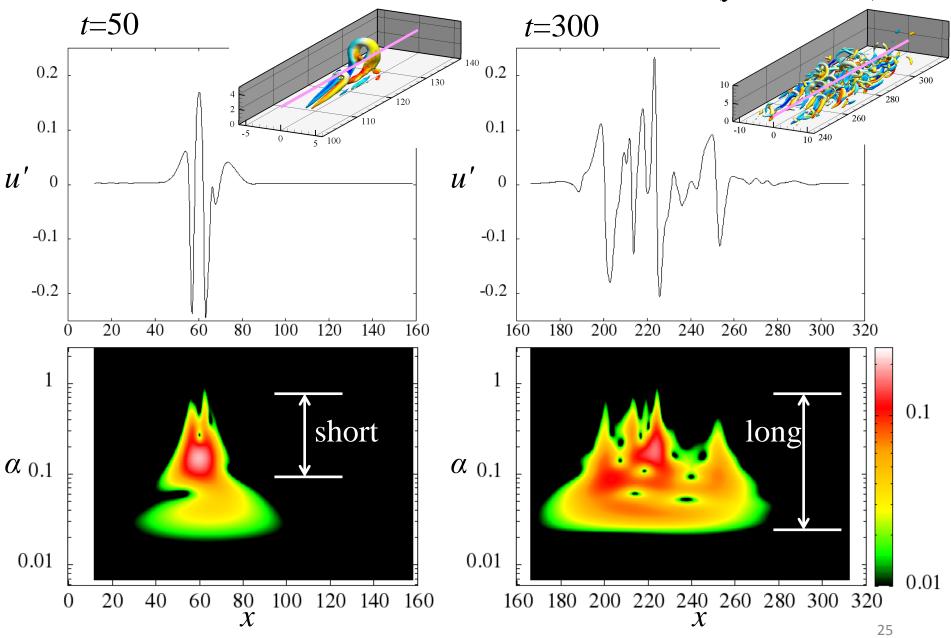
Turbulence starts around *t*=155.

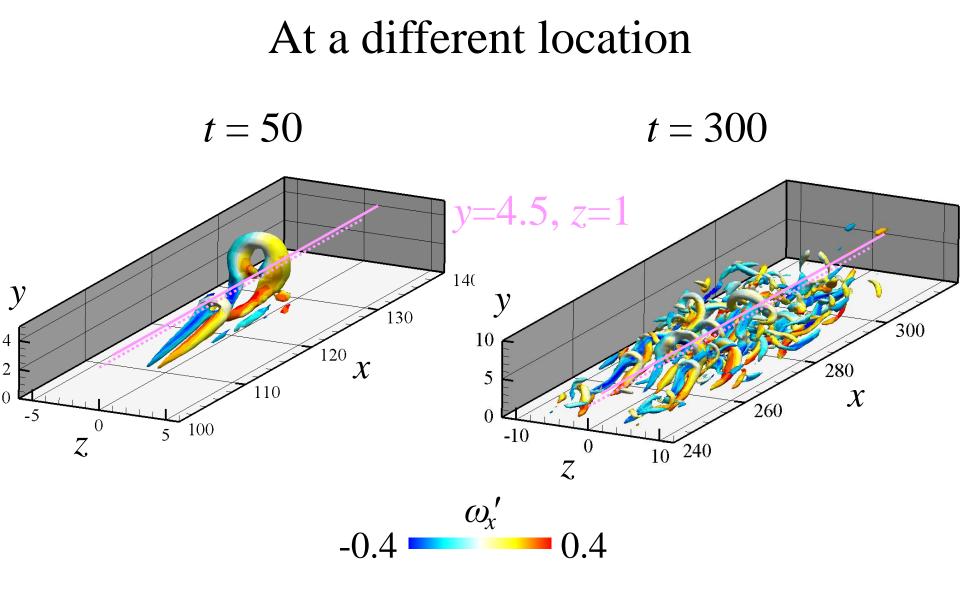
In order to identify local turbulent regions...



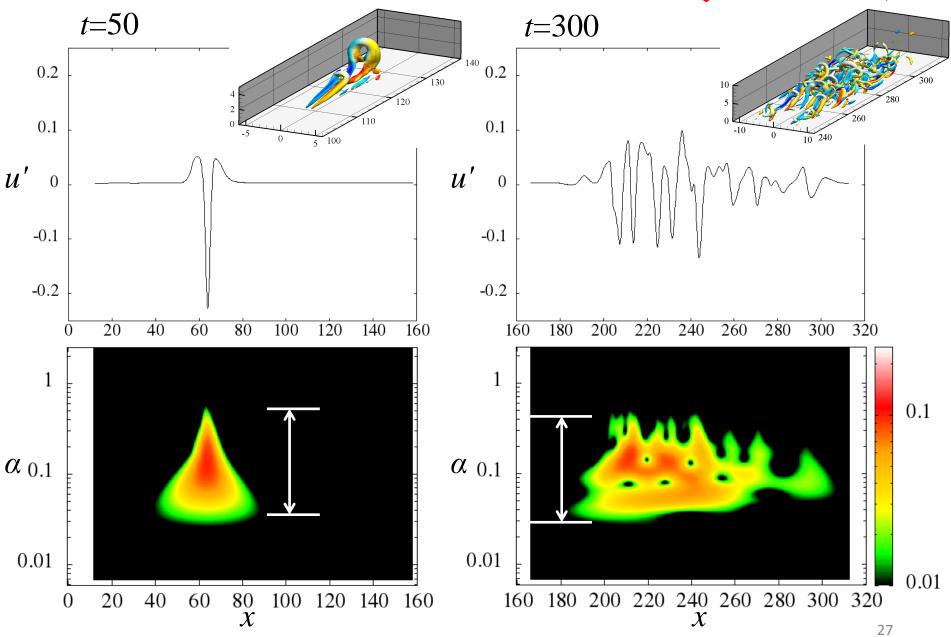
We tried to identify turbulent regions using **wavelet analysis**.

Fluctuation waves and their wavelets (y=3, z=0)





Fluctuation waves and their wavelets (y=4.5, z=1)

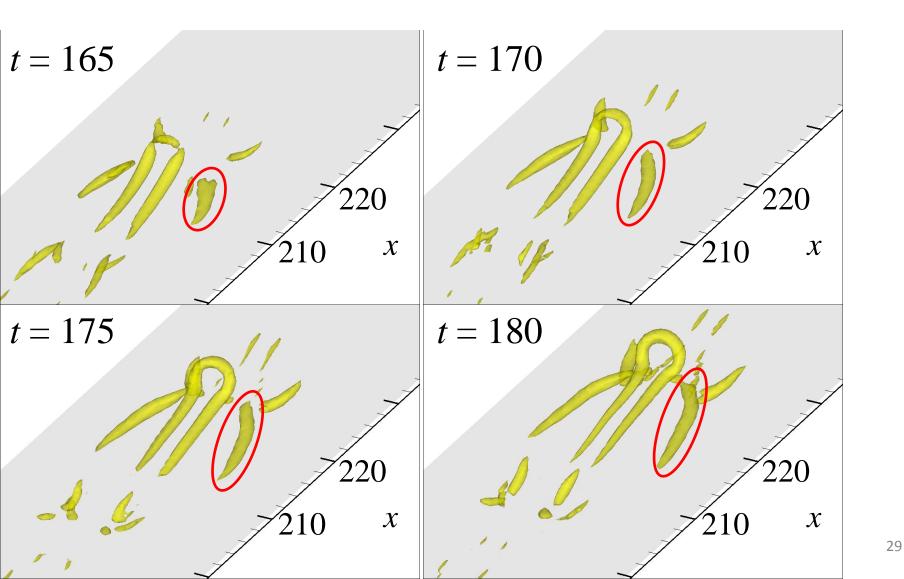


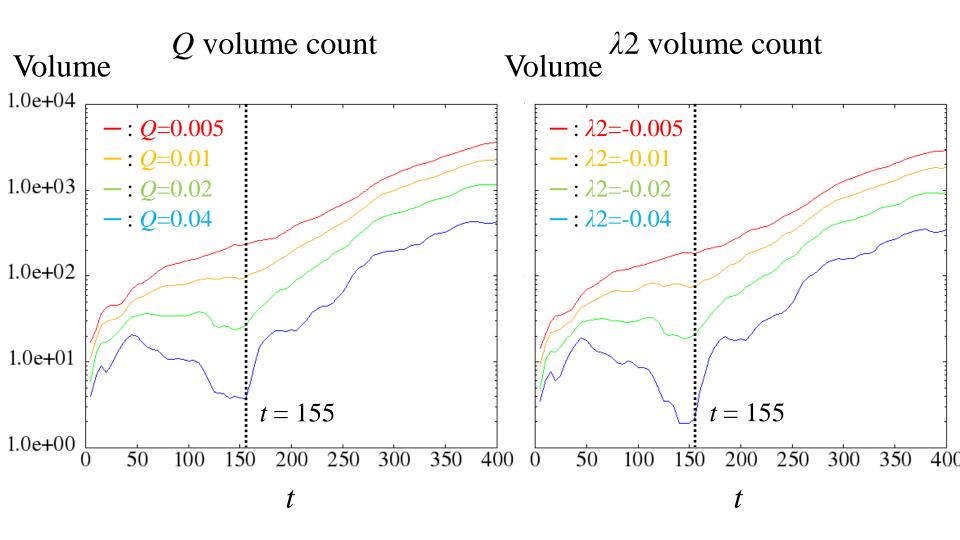
Result

Wavelet analysis failed to distinguish.

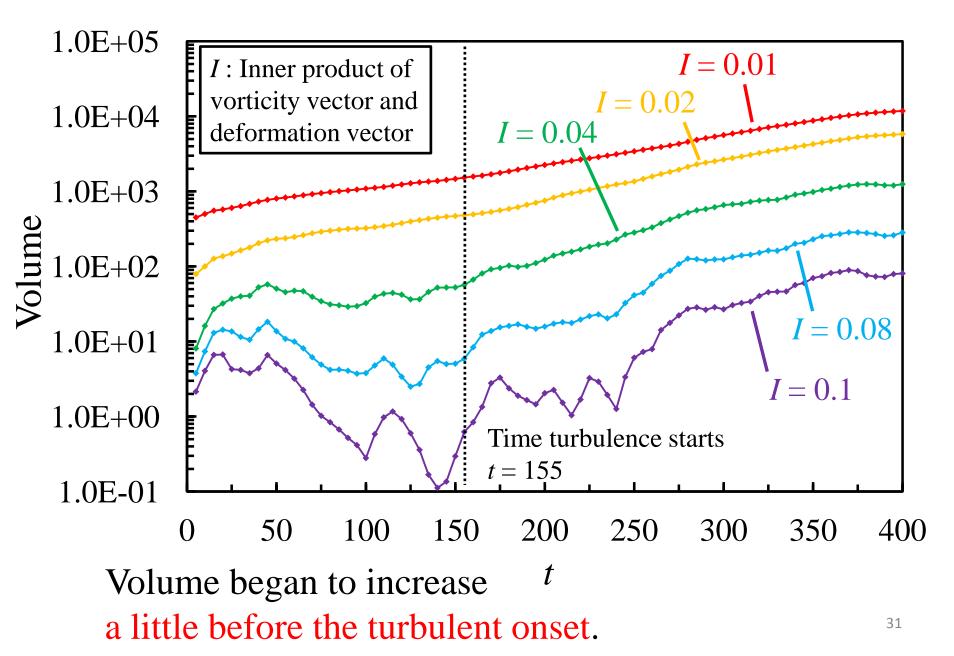
What method can?

Vortex structures soon after t=155Q=0.025





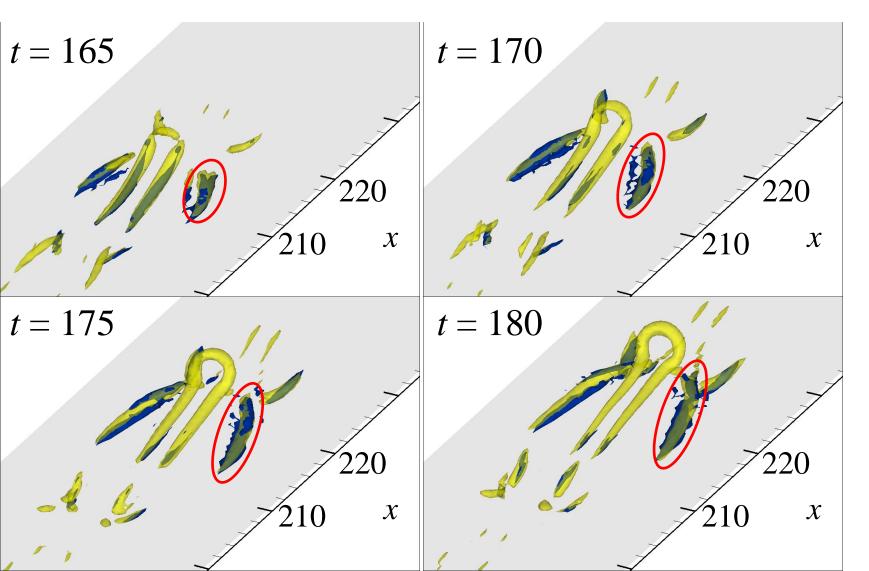
Time variation of volume of stretched region



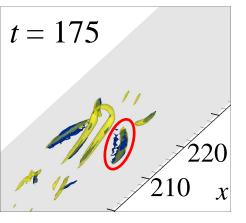
Vortex structures and stretching region Q=0.025, I=0.06 shown together

I : Inner product of vorticity vector and primary eigen vector

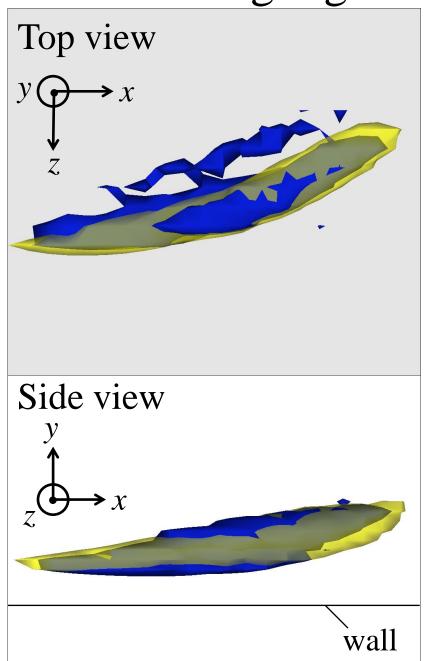
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Vortex structures and stretching region



Vortex structure



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Summary

Destabilization of flow by short duration jet introduced into the low-speed streak

was studied using 3-dimensional numerical simulation, aimed at investigating how the turbulence starts.

We tried to identify a locally turbulent region using <u>wavelet analysis</u>.

Result was:

We could not distinguish turbulent region from non-turbulent region.

We are currently trying focusing on vortex stretching.

Any suggestion is welcome!